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Article Durability of viscoelastic fibre prestressing in a polymeric composite

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Abstract: Viscoelastic fibre prestressing (VFP) is a promising technique to counterbalance the po-13 tential thermal residual stresses within a polymeric composite, offering superior mechanical bene-14 fits for structural engineering applications. It has been demonstrated that the time required for de-15 sirable creep strain can be significantly reduced by implementing higher creep stresses, while its 16 long-term reliability is still unknown. Here, we developed the prestress equivalence principle, and 17 investigated the durability of viscoelastic fibre prestressing within a composite, in order to further 18 enrich the prestress mechanisms. The effectiveness of the prestress equivalence principle was re-19 fined through Charpy impact testing of prestressed samples with various prestrain levels. The du-20 rability was investigated by subjecting samples to both natural aging (up to 0.5 years) and acceler-21 ated aging (by using the time-temperature superposition principle). It is found that the prestress 22 equivalence principle offers flexibilities for viscoelastically prestressed polymeric matrix composite 23 (VPPMC) technology; the impact benefits offered by VFP are still active after been accelerated aged 24 to an equivalent of 20,000 years at 20°C, inferring long-term reliability of VFP-generated fibre recov-25 ery within a polymeric composite. These findings demonstrated that both materials and energy con-26 sumptions could be conserved for advanced composites. Therefore, they promote further steps of 27 VPPMC technology towards potential industrial application especially for impact protections. 28

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Copyright: © 2023 by the authors. Submitted for possible open access publication under the terms and conditions of the Creative Commons Attribution (CC BY) license (https://creativecommons.org/license s/by/4.0/). Keywords: Durability; Polymeric composite; Viscoelasticity; Prestress; Impact.

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Abbreviatio	ns	31
EEST	Elastic energy storage site	32
EFP	Elastic fibre prestressing	33
EPPMC	Elastically prestressed polymeric matrix composite	34
PMC	Polymeric matrix composite	35
SE	Standard error	36
TTM	Taut-tie molecule	37
TTSP	Time-temperature superposition principle	38
VEST	Viscoelastic energy storage site	39
VFP	Viscoelastic fibre prestressing	40
VPPMC	Viscoelastically prestressed polymeric matrix composite	41
WLF	Williams-Landel-Ferry	42

1. Introduction

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Polymeric matrix composites (PMC) have been widely used in aerospace, automotive, 44 biomedical, as well as sustainable engineering [1,2]. It is known that thermal residual 45 stress levels within a PMC can be significantly affected by the fibre prestressing technique, 46 depending on the fibre and matrix combinations, as well as the laminate stacking se-47 quences [3-5]. This can be achieved through (i) elastic fibre prestressing (EFP) [6-8] and 48 (ii) viscoelastic fibre prestressing (VFP) [9,10]. For (i), tension is applied to long fibres 49 embedded in an uncured polymeric matrix; the prestress load is released on solidification 50 of the resin to produce an elastically prestressed polymeric matrix composite (EPPMC); 51 whilst for (ii), creep tension is applied to long fibres, then the load is released prior to 52 mould the fibres into a resin; following curing of the matrix, a viscoelastically prestressed 53 polymeric matrix composite (VPPMC) is manufactured [4]. 54

It has been demonstrated that both (i) and (ii) techniques can improve the mechanical 55 properties of polymeric composites without increasing their mass or structural dimen-56 sions [10], as well as reducing process-induced deformation [11]. These benefits are in-57 duced from improved internal stress levels by either EFP-generated recovery (for 58 EPPMCs) or VFP-generated recovery (for VPPMCs) along the prestressed fibres. Alt-59 hough fibre prestressing technique is promising, its applications depend on the mechani-60 cal properties of the prestressed fibres [12,13]. Commonly, EFP is applicable to brittle 61 (elastic) fibres such as carbon fibres and glass fibres [14–17], whilst VFP is more suitable 62 for tough fibres such as semi-crystalline thermoplastic fibres [18]. As for structural ap-63 plications, it is demonstrated that VPPMCs are superior than EPPMCs in terms of product 64 geometry and longevity [19,20]. To produce an EPPMC, the tension load is maintained 65 throughout the curing process which significantly restricts the product geometry, and 66 equipment designs for simultaneously stretching and moulding can be technically chal-67 lenging [21-23]; whilst for VPPMC, fibre prestressing and moulding are decoupled, 68 providing total flexibility in product geometries [24,25]. As for longevity, VFP-generated 69 fibre recovery has been demonstrated to be a long-term activity [26], while EFP-induced 70 benefits will deteriorate with time due to localised matrix creep [19,27]. Previous works 71 have shown that the VFP within a PMC can improve the tensile strength by ~15% [20], 72 impact toughness by ~50% [28–31], and flexural stiffness by ~50% [32]; the VFP-induced 73 benefits have also been applied to produce morphing (bistable) structures [33,34], green 74 composites [35], and shown great potential for viscoelastically active sutures [36]. 75

The fibre prestressing technique is effective in improving the mechanical properties 76 of PMCs, the prestress mechanisms have also been introduced. The prestressed fibre re-77 covery generates compressive stresses in the fibre/matrix interface, which in turn interact-78 ing with the thermal residual stress [37] to adjust the in-plane stress levels [38]; the fibres 79 are pre-stretched before embedded into a matrix, which is effective in destroying the de-80 fective fibres in advance, in order to reduce the energy impacts from the stress waves 81 generated by their premature failures on adjacent fibres [39,40]; the fibre prestressing pro-82 cess improves the straightness of the fibre bundles, thus increases the number of effective 83 fibres when loaded in service, and improves the load-bearing capacity of a cured compo-84 site [15]; impact of a prestressed PMC triggers the interfacial shear stress caused by fibre 85 prestressing, which promotes the interface debonding between fibre and matrix to absorb 86 energy, and thus improves the transverse impact resistance [41–43]; the compressive 87 stress generated by the prestressed fibres shifts the neutral axis of the composite under 88 bending, which interacts with the applied stress on the tension surface, hence improves 89 the flexural stiffness of the composite [32]. 90

Although it has been demonstrated that the same VFP benefits could be achieved by 91 exploiting a higher fibre creep stress over a shorter term [30], the long-term reliability of 92 the optimised VFP within a PMC is still unknown. In this research, we develop a prestress equivalence principle, and investigate the durability of viscoelastic fibre prestressing within a polymeric composite, aiming to further enrich the underlying prestress mechanisms. Since prestress benefits within a VPPMC depend on the viscoelastic recovery of the prestressed fibres, VFP levels can be represented by viscoelastic creep strain values or 97 equivalence between different creep stresses, offering further flexibilities for VPPMC technology. The durability of VFP within the PMC is then investigated through both natural aging (up to 6 months) and accelerated aging (using the time-temperature superposition principle), in order to reveal further insight into the fundamental VFP mechanisms.

2. Theoretical

2.1. Prestress equavilence principle

When a polymeric fibre is subjected to creep stress (below yield stress), it undergoes107both elastic and viscoelastic deformation. The viscoelastic behavior of creep and recov-108ery can be represented by a number of Voigt elements connected in series. Figure 1 (a)109shows the strain evolution during creep and recovery cycle of a polymeric material. The110time-dependent components represented by functions are based on the Weibull model111[44]. For creep, $\varepsilon_{ctot}(t)$ is the total strain at time t, under an applied constant creep stress:112

$$\varepsilon_{\rm ctot}(t) = \varepsilon_{\rm i} + \varepsilon_{\rm c} \left[1 - \exp\left(- \left(\frac{t}{\eta_{\rm c}} \right)^{\beta_{\rm c}} \right) \right] \tag{1}$$

Here, α is the instantaneous strain from applied stress which is time-independent; α 113 function is the time-dependent creep strain, where η_c is the characteristic life, and β_c is the shape parameter. Following removal of the creep stress, the elastic deformation is recovered, as represented by α in Figure 1 (a), and the remaining recovery strain $\alpha_{vis}(t)$ is: 116

$$\varepsilon_{\rm rvis}(t) = \varepsilon_{\rm r} \left[\exp\left(-\left(\frac{t}{\eta_{\rm r}}\right)^{\beta_{\rm r}} \right) \right] + \varepsilon_{\rm f}$$
⁽²⁾

The ε_r function is the time dependent recovery strain with η_r and β_r being the Weibull 117 parameters analogous to those in Eqn 1. The non-recoverable strain from viscous flow 118 is represented by ε_r in Figure 1 (a). 119



Figure 1. Schematic of the strain-time behaviour of a polymeric fibre under constant creep stress,121showing (a) creep-recovery strain cycle; and (b) the prestress equivalence principle.122

To get the similar prestress benefits, the same prestrain level can be achieved through 123 shorter term via higher creep stress, and the equivalence principle is illustrated in Figure 124

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1 (b). Here, fibre creep tension is applied for 24 h for convenience. Since creep strain 125 curve is represented by Eqn 1, the instantaneous strain α_1 and the time-dependent strain 126 ε (24)_{std}, can be determined experimentally under stress σ . The prestrain level is defined 127 as $[\mathcal{E}(24)_{std} - \mathcal{E}_{1}]$. To achieve the same prestrain level, the subsequent run is performed at 128 higher stress value, σ_2 (> σ_1). For strain value at $\varepsilon_c(t_n)$ to be equal to $\varepsilon_c(24)_{std}$, t_n will be < 129 24 h. Note, $\alpha(t_n)$ excludes the instantaneous strain α_2 . Therefore, a value for t_n which 130 approaches the shortest practical creep time, t_{min} , can be determined to give similar VFP 131 benefits. Thus, the same prestrain level can be achieved through shorter term via higher 132 creep stress level, and the prestress equivalence principle is mathematically: 133

$$t^{-1} = a \ln t_{\rm n} + b \tag{3}$$

where, *a* and *b* are constants depending on the prestrain level, which can be determined 134 experimentally. 135

σ

To validate the prestress equivalence principle, VPPMC samples are produced under 136 the t_n creep conditions. Thus, batches of VPPMC samples using t_n is compared with similar batches produced under standard (24 h) creep conditions to evaluate the VFP benefits. 138 Viscoelastic force, $\sigma(t)$, under t_n creep conditions is also measured and compared to the 139 standard 24 h runs, in order to provide direct experimental evidences and reveal the fundamental VFP mechanisms, which follows [45]: 136

$$\sigma(t) = \sigma_{\rm v} \left[\exp\left(-\left(\frac{\Delta t}{\eta}\right)^{\beta} \right) - \exp\left(-\left(\frac{t}{\eta}\right)^{\beta} \right) \right] \tag{4}$$

where, the σ_v function represents VFP generated time-dependent stress, as determined by the characteristic life η and shape β parameters. 143

2.2. Durability prediction through TTSP

The time-temperature superposition principle (TTSP) has been commonly used to 145 generate the master curve for tensile creep [46–48], flexural creep [49], dynamic tensile 146 modulus [50], stress relaxation [50,51], or predict the long-term viscoelastic polymeric fi-147 bre recovery [26]. The TTSP is based on the free volume theory [52,53], and adopted here 148 to investigate the durability of VFP within a polymeric composite. VPPMC samples un-149 der tn creep conditions are subjected to accelerated aging, and impact tested to determine 150 the long-term effectiveness of the VFP, to reveal the fundamental mechanisms. These are 151 achieved through transferring an elevated temperature into a time scale shift in terms of 152 free volume [54], which is mathematically expressed as [55]: 153

$$\log \frac{\eta(T)}{\eta(T_0)} = \log \alpha_{\rm T} = -\frac{B}{2.303f_0} \left(\frac{T - T_0}{f_0/a_{\rm T} + T - T_0}\right)$$
(5)

where, α_{T} is defined as the temperature shift factor; T_{0} is arbitrarily chosen as the reference 154 temperature and T is any other temperature; a_{T} is the thermal expansion coefficient of the 155 free volume fraction; f_{0} is the free volume fraction at reference temperature T_{0} . This yields 156 the well-known Williams-Landel-Ferry (WLF) equation with $C_{1}=B/(2.303f_{0})$, $C_{2}=f_{0}/a_{T}$. 157

The TTSP in Eqn 5 infers a non-linear relationship between temperature and the shift 158 factor log α_{T} ; however, nylon 6,6 fibre shows approximately linear viscoelasticity within 159 the yield creep strain, and the linear superposition principle holds [56]. This corresponds 160 with the following: (i) Howard and Williams [46] applied the TTSP to the creep of oriented 161 nylon 6,6 fibre under anhydrous conditions when a low range of creep stress (10-51 MPa) 162 was adopted, and results show that the shift factor log α T was linear to temperature; (ii) 163 Murayama et al [51] investigated the applicability of TTSP to the stress relaxation of nylon 164 6,6 fibre, and a linear relationship between shift factor log α T and temperature was ob-165 tained; (iii) similarly, a linear curve was also obtained by Dunell et al [50] through the 166 investigation into superposition of stress relaxation and dynamic tensile modulus of ny-167 lon 6,6 monofilaments at temperatures between 10 and -100°C; (iv) the linear relationship 168 was also found with unoriented nylon 6,6 filaments [57] when subjected to small creep 169

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strain values. Therefore, rather than the non-linear relation as represented in Eqn 5, a 170 simpler linear TTSP was established for oriented nylon 6,6 fibre, which is based on the 171 published data in terms of creep [46] and stress relaxation [51]. 172

The linear regression gives a gradient of 0.09765° C⁻¹[54]; this is comparable to 0.093 173 as determined by Williams and Bender [57] through the investigation of unoriented nylon 174 6,6 filaments. This enables log α_{T} to be determined at 70°C relative to 20°C, and the resulting value is -4.8825. Hence, viscoelastic activity would be 76,300 times faster at 70°C 176 relative to 20°C, *i.e.* if samples are subjected to 70°C for 2,298 h, the prestress effect from 177 viscoelastic recovery mechanisms will be aged to an equivalent of 20,000 years at 20°C, as well as to be ~25 years at 50°C ambient temperature. 179

3. Experimental

3.1. Sample preparation and durability evaluation

Composite sample preparation followed the procedures from previous studies using 182 nylon 6,6 fibres and polyester resin [31]. To concentrate on the prestress effects, a $V_{\rm f}$ of 183 ~2.0% was adopted for all composite samples and subjected to Charpy impact testing [30]. 184 Durability of VFP within the composite was evaluated through (i) naturally aged short-185 term, and middle-term impact performance, as well as (ii) long-term impact resistance 186 through accelerated aging. For (i), five batches of VPPMC samples were produced for 187 each creep condition and were either stored to 336 h (2 weeks for short-term) or 4392 h (6 188 months for middle-term) at room temperature (19-22°C), and then impact tested; for (ii), 189 samples were stored at room temperature for at least 2 weeks, and then subjected to ac-190 celerated aging prior to the impact tests. 191

A calibrated fan-assisted oven was used for accelerated aging, with a long-term tem-192 perature stability of $\pm 0.5^{\circ}$ C. Batches of samples with fibres previously subjected to creep 193 at 590 MPa for t_n h were aged, together with the standard VPPMC sample batches for 194 comparison. Owning to limitation in capacity of the fan-assisted oven, three batches of 195 VPPMC samples fabricated with each of the two creep conditions were evaluated (*i.e.* six 196 batches in total). Heat treatment was maintained at a constant 70°C for 2,298 h (3.2 197 months), which is equivalent to an exposure of 20,000 years at 20°C in terms of viscoelastic 198 recovery within the nylon fibres following the TTSP in Section 2.2. Samples were placed 199 as a single layer on the tray, and Figure 2 (a) shows the arrangements of samples before 200 aging, with individual test and control samples in alternating positions. This ensured all 201 the batches were subjected to the same heat conditions. Figure 2 (b) shows the samples 202 after accelerated aging. Samples were removed from the trays and stored in polythene 203 bags at room temperature for a further 336 h prior to impact tests. 204



Figure 2. Arrangements of samples within a tray (a) before and (b) after being subjected to the 206 accelerated aging; "T" stands for test sample, and "C" is the control (non-prestressed) counterpart. 207

3.2. Viscoelastic recovery force

Previous studies into the force output-time characteristics of viscoelastically pre-209 stressed fibres have provided useful evidences into fibre recovery [31,45]. Here, effec-210 tiveness of prestress equivalence principle is studied in terms of recovery force induced 211 by fibre prestressing within the nylon fibre, in order to provide further insight into pre-212 stress mechanisms. Procedures followed those detailed in [31]. Here, the generated re-213 covery force from up to t_n creep condition was evaluated and compared to the standard 214 24 h creep run. Since elastic deformation was fully recovered after load removal, the re-215 covery force induced by the same prestrain level, *i.e.* either from the standard 24 h or t_n h 216 (higher creep stress) run, was expected to be at similar level. 217

3.3. Statistical analysis

As stated above, multiple sample batches were tested for various optimised creep 219 conditions for repeatability, which were then compared to the standard 24 h creep run to 220 evaluate the effectiveness of the prestress equivalent principle. Thus, statistical analysis 221 is essential to evaluate the significance level of the data variances. These were performed 222 by following the standard procedures of the two-tailed hypothesis testing at a significant 223 level of 5% [58].

4. Results and discussion

4.1. Short-term effectiveness

Creep and recovery strain-time data of the nylon fibre have been characterised by 227 using the Weibull-based curve-fits from Eqn 1, the $\varepsilon_c(24)_{std}$ value (330 MPa) was found to 228 be 3.39%. Thus, for $\alpha(t_n)$ to be equal to $\alpha(24)$ _{std}, the t_n values (at ~3.4% strain level) from 229 the 395 MPa, 460 MPa, 525 MPa and 590 MPa creep data are found to be 420 min, 92 min, 230 75 min, and 37 min, respectively. Figure 3 shows the applied creep stress vs t_n values 231 with fitted logarithmic relationship following Eqn 3 at a prestrain level, $\varepsilon_c(t_n)$, of ~3.4%. 232 It offers an opportunity to predict the required creep stress for a designated t_n value. 233

The effectiveness of the trend was experimentally evaluated. This was achieved by 234 selecting a point on the extended dashed curve in Figure 3. Batches of Charpy impact 235 samples were made with the obtained creep condition, and benefits from prestress were 236 investigated through impact testing. The Eqn 3 indicates that stress values of 797 MPa 237 and 715 MPa correspond to 10 min and 15 min stretching respectively; however, initial 238 attempts on creep stress under these conditions rapidly led to fibre fracture due to stress 239 concentration effects. Since it is known that nylon fibre could sustain a 665 MPa creep 240 stress for more than 1 h, this value was adopted here, which requires a corresponding 20 241 min stretching time to achieve a prestrain level of ~3.4% in Figure 3. Thus, batches of 242 composite samples were made and Table 1 shows the impact results (tested at 336 h, *i.e.* 2 243 weeks after manufacture). The five batches of samples give an absorbed energy increase 244 of $55.78 \pm 3.77\%$. Two-sided hypothesis testing at 5% significance level shows that the 245 increase is equivalent to the benefits $(54.14 \pm 7.52\%)$ from the standard runs (330 MPa for)246 24 h) [30]. Referring to the creep strain-time data under 665 MPa [59], the strain value & 247 for $t_n = 20$ min was found to be 3.49% using Eqn 1, which is comparable to the $\varepsilon_c(24)_{std}$ 248 value at 330 MPa, i.e. 3.39%. 249

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Figure 3. Relationship between applied creep stress σ and required t_n for creep loading to get the251same prestrain level.Round data points are calculated using the prestress equivalence principle in252Figure 1 (b); the square point predicts the creep stress requirement for a stretching period of 20253minutes; curve is fitted following Eqn 3.254

These results indicate that ~3.4% viscoelastic creep strain would give an increase of 255 ~56% in absorbed impact energy. In terms of industrial application, this improvement in 256 mechanical properties could be achieved within a few minutes of fibre stretching as 257 shown in Figure 3, if stress concentration effects can be avoided. It is worth noting that 258 the prestress benefits within a composite may also be related to the stiffness ratio between 259 fibre and matrix, which infers different in-plane stress transfer mechanisms [59]. 260

	Natural	Mean impact o	Increase in energy	
Batch	age (h)	Test ± SE	Control ± SE	(%)
330 MPa (24 h)	336	33.93 ± 3.14	23.78 ± 1.48	42.64
		37.02 ± 1.78	20.71 ± 0.63	78.76
		35.36 ± 1.71	25.03 ± 0.96	41.26
		36.71 ± 2.89	25.60 ± 1.15	43.37
		35.58 ± 1.96	21.61 ± 1.13	64.69
Mean ± SE		35.7 ± 1.0	23.4 ± 0.6	54.1 ± 7.5
665 MPa (20 min)	336	34.74 ± 1.18	22.54 ± 1.16	54.14
		37.32 ± 1.44	24.00 ± 0.59	55.54
		38.42 ± 1.78	22.73 ± 0.45	68.99
		38.74 ± 2.46	25.03 ± 0.59	54.79
		34.00 ± 1.69	23.38 ± 0.55	45.46
Mean ± SE		36.65 ± 0.82 23.53 ± 0.35		55.78 ± 3.77

Table 1. Charpy impact results for VPPMC samples produced with a creep condition of 665 MPa261for 20 min; tests were performed at 336 h after manufacture; SE is the standard error.262

4.2. Recovery force equivalence

A further evidence of the prestress equivalence discussed above is to evaluate the 264 recovery force generated by the prestressed fibres. Since the optimised condition in Fig-265 ure 3 has shown effectively the same mechanical benefits for VPPMCs, the recovery force 266 with the 590 MPa for 37 min condition was investigated, and compared to the standard 267

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creep run (330 MPa for 24 h). Two runs were tested for each creep condition (for repeatability), and these were monitored up to 1000 h. Figure 4 shows the resulting data. Although there is variation in data points, the 590 MPa condition shows a slightly higher recovery rate below 10 h, which corresponds with previous findings [31]. This may due to the quick response of taut-tie molecules (TTMs) upon load removal, which would dominate for several hours. Beyond 10 h however, the 590 MPa condition shows similar recovery force values. 274



Figure 4. Recovery force measurements from nylon 6,6 fibre after being subjected to the standard276creep condition (330 MPa for 24 h) and the 590 MPa 37 min condition.277

The Weibull-based model represented by Eqn 4 was fitted to the experimental data 278 and corresponding parameter values are presented in Table 2. This enables prediction of 279 recovery force values at 336 h after creep load removal, which are expected to directly 280 relate to the absolute values of recovery force within a VPPMC at the same age. The 281 resulting data give 3.76 ± 0.06 N, and 3.46 ± 0.14 N for 330 MPa and 590 MPa conditions, 282 respectively, and two-tailed hypothesis tests show that these two forces are the same at a 283 significance level of 5%. Thus, the recovery force generated from the two creep condi-284 tions are equivalent within the experimental error. Therefore, it is concluded that recov-285 ery force from the reduced time creep (up to 590 MPa creep stress) gives a similar value 286 to the standard creep conditions at an equivalent viscoelastic creep strain level. 287

Table 2. Summary of the viscoelastic recovery force parameter values using Eqn 4; r is the288correlation coefficient.289

Creep condition -	Recovery force				
	$\sigma_{\rm v}$ (MPa)	$\varDelta t$ (h)	η (h)	β	r
330 MPa-24 h -01	5.2028	0.0414	9.6148	0.3382	0.9989
330 MPa-24 h -02	5.0470	0.1632	59.778	0.3831	0.9968
590 MPa-37 min-01	11.744	0.0676	4135.6	0.1337	0.9989
590 MPa-37 min-02	14.527	0.0751	116460	0.1548	0.9988

Although viscoelastic recovery force and short-term impact tests (336 h) provide evidences of the prestress equivalence, the long-term endurance of prestress effects does not 292

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necessarily relate to that of a VPPMC, due to the possible matrix creep effects [19]. Thus 293 the middle-term and long-term effect of VFP are further investigated and discussed below. 294

4.3. Middle-term effectiveness

Furthermore, batches of VPPMC samples with their control counterparts were pro-296 duced and allowed to age in real time to 4392 h (0.5 years). Again, batches corresponding 297 to the 590 MPa, 37 min creep condition were evaluated, together with standard 330 MPa, 298 24 h batches for comparison. Table 3 shows the Charpy impact results. Of particular 299 interest, is that the increase in impact energy in both cases is comparable, and this is veri-300 fied through a two-tailed hypothesis testing at a significance level of 5%. Therefore, the 301 viscoelastic recovery mechanism from nylon 6,6 fibre within these VPPMCs is still func-302 tional, and there is no deterioration in prestress benefits up to 0.5 years in real time aging. 303

Table 3. Summary of Charpy impact results, tested at 4392 h (0.5 years) after manufacture. Five304sample batches (5 test and 5 control samples in each batch) were tested for each viscoelastic fibre305prestressing condition; SE is standard error.306

	Natural	Mean impact	Increase in	
Batch	age (h)	Test ± SE	Control ± SE	energy (%)
330 MPa (24 h)	4392	40.00 ± 2.40	25.77 ± 0.64	55.21
		35.50 ± 3.12	20.32 ± 0.63	74.72
		37.71 ± 1.44	23.43 ± 0.81	60.96
		34.97 ± 3.71	23.13 ± 0.38	51.20
		34.31 ± 3.01	25.05 ± 1.14	36.95
Mean ± SE		36.50 ± 1.22	23.54 ± 0.47	55.81 ± 6.17
590 MPa (37 min)	4392	44.14 ± 1.82	25.17 ± 1.27	75.33
		44.26 ± 3.11	24.46 ± 0.50	80.98
		39.76 ± 2.70	25.51 ± 0.69	55.85
		35.34 ± 1.74	23.96 ± 1.78	47.49
		34.06 ± 1.90	22.90 ± 0.35	48.71
Mean ± SE		39.51 ± 1.24	24.40 ± 0.50	61.67 ± 6.94

4.4. Long-term effectiveness

When subjected to accelerated aging, apart from the sample colour (yellowish), there 308 is no significant difference visually between the sample batches, see Figure 2 (b). Table 309 4 summarises the impact results from tests two weeks after the heat treatment. By sub-310 jecting the VPPMC samples to accelerated aging, the increase in impact energy absorption 311 shows that viscoelastically generated prestress remains active for an equivalent of 20,000 312 years at a constant 20°C with both creep prestressed samples, *i.e.* as verified through a 313 two-tailed hypothesis testing (at 5% level), the mean increase value of $52.81 \pm 3.09\%$ agrees 314 with the impact result (54.14 ± 7.52%) aged for 336 h. Compared to the optimised creep 315 condition, i.e. 590 MPa for 37 min, the benefits from VFP are also similar. Hypothesis 316 testing shows that the increase of $60.43 \pm 9.52\%$ is not different to $52.81 \pm 3.09\%$ at a 5% 317 significance level. Hence, there is no deterioration in increased energy absorption. 318

Table 4. Summary of Charpy impact results.Samples were subjected to accelerated aging to an
equivalent of 20,000 years at 20°C.319with 5 test and 5 control samples in each batch; SE is standard error.321

	Exposure	Age equivalent	Mean impact	Mean impact energy (kJ m ⁻²)	
Batch	to 70°C	@ 20°C	Test ± SE	Control ± SE	energy (%)
330 MPa (24 h)	2,298	20,000	34.51 ± 2.07	22.30 ± 1.11	54.73
			33.74 ± 1.63	22.99 ± 0.77	46.76
			34.77 ± 2.32	22.15 ± 0.96	56.94

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Mean ± SE			34.34 ± 1.09	22.48 ± 0.52	52.81 ± 3.09
590 MPa (37 min)	2,298	20,000	31.78 ± 0.81	22.44 ± 0.84	41.64
			32.65 ± 0.20	18.93 ± 0.47	72.47
			35.91 ± 1.74	21.48 ± 0.33	67.18
Mean ± SE			33.45 ± 0.76	20.95 ± 0.51	60.43 ± 9.52

4.5. Durability of viscoelastic fibre prestressing

Durability of VFP within the composite is plotted in Figure 5 in terms of Charpy im-323 pact resistance. It is clear that the 590 MPa, 37 min creep conditioned VPPMC samples 324 show comparable increase in absorbed impact energy with the standard 24 h creep con-325 dition (330 MPa), in terms of both natural aging and accelerated aging. This demon-326 strates that the same prestrain level achieved through higher creep stress via a shorter 327 term remains active, and there is no deterioration in increased energy absorption in either 328 naturally aged samples up to 0.5 years or accelerated aged to an equivalent of 20,000 years 329 at room temperature. For a VPPMC produced with a viscoelastic creep strain level of 330 ~3.4%, the prestress effects can be expected not to deteriorate for at least ~25 years at a 331 constant 50°C. This would lead VPPMCs to many practical industrial applications. 332

Therefore, effects of nylon fibres prestressed through 590 MPa for 37 min is broadly 333 the same with the 330 MPa, 24 h creep condition within a composite. That is to say, VFP 334 processing time could be reduced from 24 h to tens of minutes by using higher creep stress 335 (below failure stress) with no detriment to impact performance. It is also extrapolated 336 that the logarithmic regression shown in Figure 3 is effective in long-term. Thus, it is 337 concluded that the prestress equivalence principle can be achieved by controlling the vis-338 coelastic creep strain values, i.e. the same prestrain level would generate equivalent me-339 chanical benefits within a polymeric composite. 340



Figure 5. Increase in impact energy against aging time for VPPMC samples produced under342different viscoelastic fibre prestressing conditions.343

4.6. Towards full potential of viscoelastic fibre prestressing

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It has been demonstrated that the benefits from VFP is maximised when fibres are 345 subjected to 460 MPa for 24 h, and the ε (24) value (460 MPa) is found to be 4.03% [31]. 346 Thus, further optimisation in load-time conditions could be achieved by applying the pre-347 stress equivalence principle in Section 2.1. The prestress benefits can be explored to-348 wards the full potential by applying the same principle to a higher viscoelastic creep strain, 349 *i.e.* ~4.0%. Since it is known that nylon 6,6 can sustain a 665 MPa creep stress for more 350 than 1 h, then for ε_{tn} equals to $\varepsilon_{t}(24)$ at 460 MPa, the t_n values for creep at 590 MPa and 351 665 MPa were found to be 134 min and 48 min, respectively. Data from corresponding 352 Charpy impact results are summarised in Table 5. For reference, batches subjected to 460 353 MPa for 24 h from Ref [31] are also presented. 354

The resulting mean data from Table 5 are shown in Figure 6 for comparison. Again, 355 the optimised conditions are in good agreement with the 24 h creep runs in terms of in-356 crease in impact energy absorption. Although there is a slight decrease in increased im-357 pact energy over the three creep conditions, from 79% to 73%, little difference was ob-358 served in impact energy absorption (in absolute terms). In fact, two-tailed hypothesis 359 testing results show that there is no significant difference in absorbed energy increase, *i.e.* 360 they are similar. Compared to the creep strain level at ~3.4%, raising the viscoelastic 361 creep strain level to ~4.0% gives a mean increase in energy absorption of ~75%. Thus, an 362 18% increase in prestrain level results in a further increase of ~34% in prestress benefits. 363

Table 5. Summary of Charpy impact test results, tested at 336 h and 4392 h after manufacture. Five364sample batches were tested for each prestressing condition with 5 test and 5 control samples in each365batch; SE is standard error.366

	Natural age	Mean impact	Increase in	
Batch	(h)	Test ± SE Control ± SE		energy (%)
460 MPa (24 h)	336	39.63 ± 2.22	24.08 ± 0.74	64.60
		37.30 ± 0.54	22.28 ± 0.88	67.39
		38.57 ± 1.35	19.46 ± 0.15	98.21
		39.36 ± 0.95	22.59 ± 0.66	74.23
		44.81 ± 1.86	23.41 ± 0.59	91.36
Mean ± SE		39.93 ± 0.81	22.36 ± 0.42	79.16 ± 6.66
590 MPa (134 min)	336	38.73 ± 1.76	20.79 ± 0.62	86.26
		39.38 ± 1.19	23.18 ± 0.68	69.90
		38.35 ± 1.95	23.70 ± 0.51	61.82
		42.00 ± 3.15	23.25 ± 0.74	80.69
		41.75 ± 3.39	24.06 ± 1.49	73.53
Mean ± SE		40.04 ± 0.92	22.99 ± 0.35	74.44 ± 4.24
665 MPa (48 min)	336	42.07 ± 4.99	23.91 ± 0.49	75.95
		43.29 ± 1.86	23.87 ± 0.66	81.23
		37.57 ± 1.53	22.81 ± 0.60	64.67
		39.29 ± 2.10	23.66 ± 0.72	66.05
		44.07 ± 2.84	25.16 ± 0.09	75.17
Mean ± SE		41.26 ± 1.30	23.89 ± 0.28	72.61 ± 3.15
590 MPa (134 min)	4392	37.65 ± 2.09	26.05 ± 0.74	44.54
		38.57 ± 2.44	25.14 ± 0.63	53.38
		36.16 ± 3.16	22.82 ± 0.46	58.48
		40.19 ± 3.49	22.09 ± 0.66	81.95
		42.26 ± 4.60	23.20 ± 0.28	82.17
Mean ± SE		38.96 ± 1.21	23.86 ± 0.42	64.10 ± 7.66

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0

460 MPa (24 h)

Impact energy absorbed (kJm⁻²)



590 MPa (134 min)

Figure 6. Charpy impact test results, the three creep conditions are equivalent in terms of final369viscoelastic creep strain for &(24) at 460 MPa, *i.e.* ~4.0%; error bars represent the standard error.370

665 MPa (48 min)

A further five batches of VPPMC samples with their control counterparts, produced 371 with the 590 MPa, 134 min creep condition were stored for 4392 h (0.5 years) prior to impact testing to investigate stability in prestress benefits. The results are also shown in 373 Table 5. Again, all batches show effectively the same mechanical benefits from VFP (at 374 5% significance level). Therefore, it is concluded that there is no deterioration in impact 375 performance of VPPMC samples at ~4.0% creep strain level for up to 0.5 years in real time. 376

4.7. Viscoelastic fibre prestressing mechanisms

Figure 7 shows the relationship between applied stress σ and t_n at a prestrain level of 378 ~4.0% following the principle described in Figure 1, together with the curve at ~3.4% prestrain level from Figure 3, and fitted by Eqn 3. It shows that the two curve-fits meet at 380 ~0.1 h with a creep stress of ~920 MPa, which infers that the fibre breaks after very short 381 time loading. Since the breaking strength of nylon fibre is up to 900 MPa, again, Eqn 3 indicates reasonable predictions on prestress equivalence. 383

Here, two viscoelastic creep strain levels were adopted, where a ~3.4% (Section 4.1) 384 prestrain level represents the standard creep condition (330 MPa for 24 h), and ~4.0% rep-385 resents the strain when the maximum mechanical benefits were observed for all the pre-386 stress conditions under investigation. For ~3.4%, the impact benefits from the optimised 387 VPPMC samples are broadly the same with the standard prestressed samples (330 MPa 388 for 24 h). Also, there is no detriment to impact behaviour in samples either naturally 389 aged to 0.5 years or artificially aged to an equivalent (in terms of TTSP) of 20,000 years at 390 a constant 20°C. This infers the long-term reliability of VFP in a PMC. Confirmation of 391 effectiveness is also demonstrated at a viscoelastic creep strain (prestrain) of ~4.0%, where 392 prestress equivalence is observed from the impact performance of VPPMC samples natu-393 rally aged up to 0.5 years in real time. The effect of viscoelastic creep strain level has 394 shown a decrease in benefits when fibres are subjected to 590 MPa for 24 h, i.e. ~4.8% in 395 viscoelastic creep strain [31]. These findings further developed the understanding of the 396 viscoelastic prestress mechanisms within the composite. 397

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Figure 7. Relationship between applied creep stress σ and required t_n for loading at two viscoelastic 399 creep strain levels; dashed lines are fitted by using Eqn 3. 400

The VFP mechanisms are closely related to the microstructures of the prestressed fi-401 bres. For a viscoelastic solid, the time-dependent deformation can be represented by a 402 series of springs and dashpots. These are treated as consist of elastic energy storage sites 403 (EESTs) and viscoelastic energy storage sites (VESTs) [59]. For a semi-crystalline poly-404meric fibre, instantaneous elastic deformation is mainly determined by the crystalline re-405 gions, i.e. EESTs; whilst viscoelastic deformation is dominated by the amorphous regions, 406 *i.e.* VESTs. As implied by the free volume theory and the TTSP, various prestrain levels 407 can be achieved under the same creep stress for different durations. Figure 8 shows the 408 increase in impact energy under a creep stress of 590 MPa, versus the VFP processing time. 409 When subjected to a constant creep stress, VESTs are progressively triggered to store en-410 ergy. As the creep stretching progresses over longer periods, an optimum level for pre-411 stress benefits from stored energy releasing was observed. Thus effectively, increasing 412 the creep stress or increasing the creep time has similar effects on VESTs, which controls 413 the mechanical performance of VPPMCs, as demonstrated in Figure 7. This also provides 414a theoretical foundation for the fact that time-stress superposition principle could be well 415 fitted to the strain-time data of nylon 6,6 fibre [47], i.e. stress effects can be represented by 416 time duration at a constant stress value. 417



Figure 8. Summary of increase in impact energy from VPPMC samples under a creep stress of 590419MPa. Viscoelastic creep strain levels are achieved through creep exposure, *i.e.* 37 min for ~3.4%,420134 min for ~4.0% and 24 h for 4.8%.421

5. Conclusions

This research develops the prestress equivalence principle, and investigated the du-423 rability of viscoelastic fibre prestressing within a composite, in order to achieve towards 424 the full potential of prestress benefits for VPPMC technology, and further enrich the pre-425 stress mechanisms. The effectiveness of the prestress equivalence principle was refined 426 through Charpy impact testing of prestressed samples with different prestrain levels; du-427 rability was analysed by subjecting VPPMC samples to both natural aging (up to 0.5 years) 428 and accelerated aging (by applying the time-temperature superposition principle). The 429 VFP mechanisms are then proposed. The main findings include: 430

(i) The durability of viscoelastic fibre prestressing in a polymeric composite is
evaluated through both natural aging (up to 0.5 years) and accelerated aging. It is found
that the impact benefits are still active after been accelerated aged to an equivalent of
20,000 years at 20°C, inferring long-term reliability of VFP-generated fibre recovery as in
a composite.

(ii) The developed prestress equivalence principle shows a logarithmic relationship between applied creep stress σ and t_n , allowing prediction of the t_n value required for a given σ . This is further verified by exploiting various creep stress conditions to obtain the same prestrain level. It can also be applied to pursue towards achieving the full mechanical potential of the viscoelastic fibre prestressing, indicating further flexibilities for VPPMC technology. 436

(iii) Longer exposure of nylon 6,6 yarns to a higher strain level could increase the
viscoelastic prestress-induced mechanical benefits. Compared to the ~3.4% prestrain
level, an 18% increase in viscoelastic creep strain results in a ~34% increase in prestress
benefits, and there is no deterioration in prestress benefits up to 0.5 years at real time.

(iv) The increase in impact energy is a function of creep time at a constant creep
stress (590 MPa). There is an optimum prestrain level to maximise the prestress benefits.
Increasing the creep stress (at constant creep time) or the creep time (at constant creep
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stress) have similar effects on the microstructures of the prestressed fibres; the prestress 449 mechanisms are subsequently proposed. 450

These findings demonstrated that both materials and energy consumptions could be451conserved for manufacturing of advanced composites, the viscoelastic fibre prestressing452presents long-term effectiveness within a polymeric composite.Therefore, they promotefurther steps of VPPMC technology towards potential industrial applications, especially454for impact protections.455

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Refer	rences	467
1.	Wang, B.; Zhong, S.; Lee, TL.; Fancey, K.S.; Mi, J. Non-Destructive Testing and Evaluation of Composite	468
	Materials/Structures: A State-of-the-Art Review. Adv. Mech. Eng. 2020, 12, 1–28, doi:10.1177/1687814020913761.	469
2.	Mousavi, S.M.; Hashemi, S.A.; Kalashgrani, M.Y.; Omidifar, N.; Bahrani, S.; Vijayakameswara Rao, N.; Babapoor, A.;	470
	Gholami, A.; Chiang, WH. Bioactive Graphene Quantum Dots Based Polymer Composite for Biomedical Applications.	471
	Polymers (Basel). 2022, 14.	472
3.	Hull, D.; Clyne, T.W. An Introduction to Composite Materials; Cambridge University Press, 1996; ISBN 0521388554.	473
4.	Fancey, K.S. Viscoelastically Prestressed Polymeric Matrix Composites: An Overview. J. Reinf. Plast. Compos. 2016, 35, 1290-	474
	1301, doi:10.1177/0731684416649036.	475
5.	Daynes, S.; Weaver, P.M. Stiffness Tailoring Using Prestress in Adaptive Composite Structures. <i>Compos. Struct.</i> 2013 , <i>106</i> , 282–287	476 477
6	Zor 201. Zhigun I.G. Experimental Evaluation of the Effect of Prestressing the Fibers in Two Directions on Certain Elastic	478
0.	Characteristic of Woven-Class Reinforced Plastics Polum Mech 1968 4 691–695 doi:10.1007/BE00855802	479
7	Tuttle M.F. A. Mechanical/Thermal Analysis of Prestressed Composite Laminates I. Compos. Mater. 1988, 22, 780–792	480
7.	doi:10.1177/002199838802200806	481
8	Zaidi B.M. Magniez, K. Miao, M. Prestressed Natural Fibre Spun Yarn Reinforced Polymer-Matrix Composites. <i>Compos</i>	482
0.	Part A Ann. Sci Manuf 2015 75 68–76 doi:https://doi.org/10.1016/i.compositesa.2015.04.021	483
9	Fancey, K.S. Investigation into the Feasibility of Viscoelastically Generated Pre-Stress in Polymeric Matrix Composites	484
	Mater. Sci. Eng. A 2000, 279, 36–41, doi:10.1016/S0921-5093(99)00687-5	485
10.	Ogunleve, R.O.: Rusnakova, S. A Review of Prestressed Fibre-Reinforced Polymer Matrix Composites, <i>Polymers (Basel)</i> .	486
101	2022, 14.	487
11.	Liu, Z.; Zheng, X.; Fan, W.; Wang, F.; Ahmed, S.; Yan, L. An Alternative Method to Reduce Process-Induced Deformation	488
	of CFRP by Introducing Prestresses. Chinese J. Aeronaut. 2022, 35, 314–323, doi:https://doi.org/10.1016/j.cja.2022.03.005.	489
12.	Chen, H.; Yu, F.; Wang, B.; Zhao, C.; Chen, X.; Nsengiyumva, W.; Zhong, S. Elastic Fibre Prestressing Mechanics within a	490
	Polymeric Matrix Composite. Polymers (Basel). 2023, 15.	491
13.	Jokūbaitis, A.; Valivonis, J. An Analysis of the Transfer Lengths of Different Types of Prestressed Fiber-Reinforced Polymer Reinforcement. <i>Polymers (Basel)</i> . 2022, 14.	492 493
14.	Mostafa, N.H.: Ismarrubie, Z.N.: Sapuan, S.M.: Sultan, M.T.H. Fibre Prestressed Composites: Theoretical and Numerical	494
	Modelling of Unidirectional and Plain-Weave Fibre Reinforcement Forms, Compos. Struct, 2017, 159, 410–423,	495
	doi:10.1016/j.compstruct.2016.09.090.	496
15.	Mohamed, M.; Selim, M.M.; Ning, H.; Pillay, S. Effect of Fiber Prestressing on Mechanical Properties of Glass Fiber Epoxy	497
	Composites Manufactured by Vacuum-Assisted Resin Transfer Molding. J. Reinf. Plast. Compos. 2020, 39, 21–30,	498
	doi:10.1177/0731684419868841.	499
16.	Chillara, V.S.C.; Dapino, M.J. Mechanically-Prestressed Bistable Composite Laminates with Weakly Coupled Equilibrium	500
	Shapes. <i>Compos. Part B Eng.</i> 2017 , <i>111</i> , 251–260, doi:10.1016/j.compositesb.2016.12.011.	501
17.	Mohamed, M.; Brahma, S.; Ning, H.; Pillay, S. Monitoring of Mechanical Properties of Prestressed Glass Fiber Epoxy	502
	Composites over 12 Months after Fabrication. J. Compos. Mater. 2021 , 55, 3001–3011, doi:10.1177/00219983211004706.	503
18.	Mostafa, N.H.; Ismarrubie, Z.N.; Sapuan, S.M.; Sultan, M.T.H. Fibre Prestressed Polymer-Matrix Composites: A Review. J.	504
	Compos. Mater. 2016, doi:10.1177/0021998316637906.	505
19.	Fancey, K.S. Fiber-Reinforced Polymeric Composites with Viscoelastically Induced Prestress. J. Adv. Mater. 2005, 37, 21–29.	506
20.	Pang, J.W.C.; Fancey, K.S. Analysis of the Tensile Behaviour of Viscoelastically Prestressed Polymeric Matrix Composites.	507
	<i>Compos. Sci. Technol.</i> 2008 , <i>68</i> , 1903–1910, doi:10.1016/j.compscitech.2007.12.018.	508

21.	Krishnamurthy, S. Pre-Stressed Advanced Fibre Reinforced Composites Fabrication and Mechanical Performance, Cranfield University, 2006.	509 510
22	Davnes, S.; Diaconu, C.G.; Potter, K.D.; Weaver, P.M. Bistable Prestressed Symmetric Laminates, J. Compos. Mater. 2010, 44.	511
	1119–1137, doi:10.1177/0021998309351603.	512
23.	Mostafa, N.H.; Ismarrubie, Z.N.; Sapuan, S.M.; Sultan, M.T.H. Effect of Fabric Biaxial Prestress on the Fatigue of Woven E-	513
	Glass/Polvester Composites. Mater. Des. 2016 , 92, 579–589, doi:http://dx.doi.org/10.1016/i.matdes.2015.12.109.	514
24.	Fazal, A.; Fancey, K.S. Viscoelastically Prestressed Polymeric Matrix Composites–Effects of Test Span and Fibre Volume	515
	Fraction on Charpy Impact Characteristics, Compos. Part B Eng. 2013, 44, 472–479, doi:10.1016/i.compositesb.2012.04.004.	516
25.	Oin, Y.: Fancey, K.S. Viscoelastically Prestressed Polymeric Matrix Composites – Effects of Delayed Moulding on Charpy	517
	Impact Properties Compos. Part A Appl. Sci. Manuf. 2019, 121, 169–174, doi:https://doi.org/10.1016/i.compositesa.2019.03.014	518
26.	Pang, I W C : Fancey, K S. An Investigation into the Long-Term Viscoelastic Recovery of Nylon 6.6 Fibres through	519
20.	Accelerated Ageing Mater Sci Fug A 2006 431 100–105 doi:10.1016/j.msea.2006.05.052	520
27	Mostafa N H · Ismarrubie Z N · Sanuan S M · Sultan M T H The Influence of Equi-Biavially Fabric Prestressing on the	520
27.	Flavural Parformance of Woven E-Class/Polyester-Reinforced Composites I Common Mater 2016 50 3385-3393	521
	doi:10.1177/0021998315620478.	523
28	Fazal A : Fancey K S LIHMWPE Fibre-Based Composites: Prestress-Induced Enhancement of Impact Properties. <i>Compos</i>	524
20.	Part B Eng 2014 66 1–6 doi:10.1016/i.compositesb 2014.04.031	525
29	Fancey K.S. Viscoelastically Prestressed Polymeric Matrix Composites – Potential for Useful Life and Impact Protection	526
27.	Compos Part B Eng. 2010 41 454–461 doi:10.1016/i compositesb 2010.05.002	527
30	Wang B: Fancey K S. Towards Ontimisation of Load-Time Conditions for Producing Viscoalastically Prestressed	528
50.	Polymeric Matrix Composites Courses Part B Eng. 2016 87, 336–342, doi:10.1016/j.compositesb.2015.09.003	520
31	Wang B: Fancey, K.S. Viscoelastically Prostressed Polymeric Matrix Composites: An Investigation into Fibre Deformation	520
51.	and Prostross Machanisms, Commos Part A Ann. Sci. Manuf 2018, 111, 106, 114, doi:10.1016/j.compositoca.2018.05.013	521
30	Pang LWC: Eancow KS. The Elevural Stiffness Characteristics of Viccoalastically Prostrossed Polymeric Matrix	522
32.	Compositos Courses Part A Anal. Sci. Maruf 2000, 40, 784, 700, doi:10.1016/j.compositos. 2000.02.000	552
22	Composites. Compos. Part A Appl. Sci. Manuf. 2009, 40, 784–790, doi:10.1016/j.compositesa.2009.03.009.	533
33.	Wang, B.; Fancey, K.S. A Bistable Morphing Composite Using Viscoelastically Generated Prestress. <i>Mater. Lett.</i> 2015, 158, 108–110. doi:10.1016/j.matlet.2015.05.129	534
34	Wang B: Ce. C: Fancey, K.S. Snan-through Behaviour of a Bistable Structure Based on Viscoelastically Cenerated	536
54.	Prostross Courses Part B Eng 2017 114 22–33 doi:10.1016/j.compositesb.2017.01.069	537
35	Oin V : Fancey, K.S. Towards "Creen" Viscoelastically Prestressed Composites: Cellulose Fibre Reinforcement Courses	538
55.	Dart P. Erg. 2018, 154, 420, 448, doi:10.1016/j.compositesb.2018.08.006	530
26	Fundo L.A. Engery K.S. Vicconlastically Active Sutures A Stitch in Time? Mater Sci. Ere C 2021, 121, 111605	539
30.	dei:10.1016/j.marc. 2020.111605	540
07		541
37.	Wisnom, M.K.; Gigliotti, M.; Ersoy, N.; Campbell, M.; Potter, K.D. Mechanisms Generating Residual Stresses and Distortion	542
	during Manufacture of Polymer–Matrix Composite Structures. Compos. Part A Appl. Sci. Manuf. 2006, 37, 522–529,	543
• •	doi:https://doi.org/10.1016/j.compositesa.2005.05.019.	544
38.	Tuttle, M.E.; Koehler, R.T.; Keren, D. Controlling Thermal Stresses in Composites by Means of Fiber Prestress. J. Compos.	545
	Mater. 1996, 30, 486–502, doi:10.1177/002199839603000404.	546
39.	Mills G J; Dauksys, R.J. Effects of Prestressing Boron/Epoxy Prepreg on Composite Strength Properties. AIAA J. 1973, 11,	547
	1459–1460, doi:10.1016/b978-0-08-011660-0.50015-5.	548
40.	Manders, P.W.; Chou, T.W. Enchancement of Strength in Composites Reinforced with Previously Stressed Fibers. J.	549
	<i>Compos. Mater.</i> 1983 , 17, 26–44, doi:10.1177/002199838301700103.	550

41.	Fancey, K.S. Viscoelastically Prestressed Polymeric Matrix Composites – Potential for Useful Life and Impact Protection.	551
	Compos. Part B 2010, 41, 454–461, doi:10.1016/j.compositesb.2010.05.002.	552
42.	Fazal, A.; Fancey, K.S. Viscoelastically Prestressed Polymeric Matrix Composites – Effects of Test Span and Fibre Volume	553
	Fraction on Charpy Impact Characteristics. Compos. Part B 2013, 44, 472-479, doi:10.1016/j.compositesb.2012.04.004.	554
43.	Fazal, A.; Fancey, K.S. UHMWPE Fibre-Based Composites : Prestress-Induced Enhancement of Impact Properties. Compos.	555
	Part B Eng. 2014, 66, 1–6, doi:10.1016/j.compositesb.2014.04.031.	556
44.	Fancey, K.S. A Latch-Based Weibull Model for Polymerie Creep and Recovery. J. Polym. Eng. 2001, 21, 489–510,	557
	doi:10.1515/POLYENG.2001.21.6.489.	558
45.	Pang, J.W.C.; Lamin, B.M.; Fancey, K.S. Force Measurement from Viscoelastically Recovering Nylon 6, 6 Fibres. Mater. Lett.	559
	2008 , <i>62</i> , 1693–1696, doi:10.1016/j.matlet.2007.09.061.	560
46.	Howard, W.H.; Williams, M.L. The Viscoelastic Properties of Oriented Nylon 66 Fibers: Part I: Creep at Low Loads and	561
	Anhydrous Conditions. Text. Res. J. 1963, 33, 689–696, doi:10.1177/004051756303300903.	562
47.	Wang, B.; Fancey, K.S. Application of Time-Stress Superposition to Viscoelastic Behavior of Polyamide 6,6 Fiber and Its	563
	"true" Elastic Modulus. J. Appl. Polym. Sci. 2017, 134, 1–9, doi:10.1002/app.44971.	564
48.	Fairhurst, A.; Thommen, M.; Rytka, C. Comparison of Short and Long Term Creep Testing in High Performance Polymers.	565
	Polym. Test. 2019, 78, 105979, doi:https://doi.org/10.1016/j.polymertesting.2019.105979.	566
49.	Chevali, V.S.; Dean, D.R.; Janowski, G.M. Flexural Creep Behavior of Discontinuous Thermoplastic Composites: Non-	567
	Linear Viscoelastic Modeling and Time-Temperature-Stress Superposition. Compos. Part A Appl. Sci. Manuf. 2009, 40, 870-	568
	877, doi:10.1016/j.compositesa.2009.04.012.	569
50.	Dunell, B.A.; Joanes, A.A.; Rye, R.T.B. Viscoelastic Behavior of Nylon 6-6 Monofilaments below Room Temperature. J.	570
	Colloid Sci. 1960, 15, 193–204, doi:10.1016/0095-8522(60)90021-0.	571
51.	Murayama, T.; Dumbleton, J.H.; Williams, M.L. The Viscoelastic Properties of Oriented Nylon 66 Fibers. Part III: Stress	572
	Relaxation and Dynamic Mechanical Properties. J. Macromol. Sci. Part B Phys. 1967, 1, 1–14, doi:10.1080/00222346708212736.	573
52.	Doolittle, A.K. Studies in Newtonian Flow. II. The Dependence of the Viscosity of Liquids on Free-space. J. Appl. Phys. 1951,	574
	22, 1471–1475.	575
53.	Starkova, O.; Gagani, A.I.; Karl, C.W.; Rocha, I.B.C.M.; Burlakovs, J.; Krauklis, A.E. Modelling of Environmental Ageing of	576
	Polymers and Polymer Composites— Durability Prediction Methods. Polymers (Basel). 2022, 14.	577
54.	Fancey, K.S.; Fazal, A. Prestressed Polymeric Matrix Composites: Longevity Aspects. Polym. Compos. 2016, 37, 2092–2097,	578
	doi:10.1002/pc.23387.	579
55.	Ferry, J.D. Viscoelastic Properties of Polymers; John Wiley & Sons, 1980; ISBN 0471048941.	580
56.	Findley, W.N.; Davis, F.A. Creep and Relaxation of Nonlinear Viscoelastic Materials. Courier Corporation 2013.	581
57.	Williams, M.L.; Bender, M.F. Extension of Unoriented Nylon 66 Filaments. III. Superposition of Data. J. Appl. Phys. 1965, 36,	582
	3044–3049, doi:10.1063/1.1702925.	583
58.	Devore, J.L. Probability and Statistics for Engineering and the Sciences; Cengage learning, 2011; ISBN 0538733527.	584
59.	Wang, B. Viscoelastically Prestressed Composites: Towards Process Optimisation and Application to Morphing Structures,	585
	University of Hull, 2016.	586
		587
		399